## Dynamics - Rigid Body Dynamics

## Introduction

- A rigid body is a many-particle system in which the distance between particles is fixed. The location of all particles is described by $\mathbf{6}$ coordinates - 3 spatial and 3 angular.
- The velocity is determined by $\boldsymbol{v}$, the velocity of the $\mathbf{C o M}$ and $\boldsymbol{\omega}$, the angular velocity.
- The basic two equations of angular motion are

$$
M \ddot{\boldsymbol{R}}=\boldsymbol{F}_{0}
$$

The centre of mass moves as if it were a single particle under the action of a force $\boldsymbol{F}_{\boldsymbol{0}}$.

$$
\dot{\boldsymbol{J}}=\boldsymbol{G}_{0}
$$

The rate of change of angular momentum is equal to the total applied couple.

- Other basic equations:
o The velocity $\boldsymbol{v}$ of a particle at a distance $\boldsymbol{r}$ from an axis around which a rotation at speed $\boldsymbol{\omega}$ is happening is

$$
\boldsymbol{v}=\boldsymbol{\omega} \times \boldsymbol{r}
$$

o For similar reasons:

$$
\frac{\mathrm{d} \boldsymbol{J}}{\mathrm{~d} t}=\boldsymbol{\omega} \times \boldsymbol{J}
$$

o Angular speeds are additive. To if frame 1 is rotating with $\boldsymbol{\omega}_{1 \text { wrt } 2}$ with respect to frame 2, which is rotating with $\boldsymbol{\omega}_{2 \text { wrt } 3}$ with respect o frame 3, then

$$
\boldsymbol{\omega}_{1 \mathrm{wrt} 3}=\boldsymbol{\omega}_{1 \mathrm{wrt} 2}+\boldsymbol{\omega}_{2 \mathrm{wrt} 3}
$$

## Relating $\boldsymbol{J}$ and $\boldsymbol{\omega}$

- If the body is rotating at $\boldsymbol{\omega}$, the total angular momentum is given by

$$
\begin{aligned}
\boldsymbol{J} & =\sum \boldsymbol{r} \times \boldsymbol{p} \\
& =\sum \boldsymbol{r} \times m(\boldsymbol{\omega} \times \boldsymbol{r}) \\
& =\sum m\left[r^{2} \boldsymbol{\omega}-(\boldsymbol{\omega} \cdot \boldsymbol{r}) \boldsymbol{r}\right] \\
& =\sum m\left[r^{2} \boldsymbol{\omega}-\left(\omega_{x} x+\omega_{y} y+\omega_{z} z\right) \boldsymbol{r}\right]
\end{aligned}
$$

In detail

$$
\boldsymbol{J}=\underbrace{\left.\begin{array}{ccc}
\sum m\left(y^{2}+z^{2}\right) & -\sum m x y & -\sum m x z \\
-\sum m x y & \sum m\left(x^{2}+z^{2}\right) & -\sum m y z \\
-\sum m x z & -\sum m y z & \sum m\left(x^{2}+y^{2}\right)
\end{array}\right) \boldsymbol{\omega}}_{\boldsymbol{I}} \underset{\boldsymbol{J}=\boldsymbol{I} \boldsymbol{\omega}}{ }
$$

[The non-diagonal elements are fairly easy to derive. The diagonal ones should actually have $x^{2}+y^{2}+z^{2}$, because one of the terms is always knocked out by the second term in the sum]. In other words, $\boldsymbol{J}$ is proportional to $\boldsymbol{\omega}$, but not necessarily parallel to it.

- The off-axes elements are rather hard to understand - they correspond to the fact that looking at a particle at a given instant, it's impossible to tell exactly around which axis it's moving.
- Also, we can find the kinetic energy

$$
\begin{gathered}
T=\sum \frac{1}{2} m[(\boldsymbol{\omega} \times \boldsymbol{r}) \cdot(\boldsymbol{\omega} \times \boldsymbol{r})] \\
=\sum \frac{1}{2} m[\boldsymbol{\omega} \cdot \boldsymbol{r} \times(\boldsymbol{\omega} \times \boldsymbol{r})] \\
T=\frac{1}{2} \boldsymbol{\omega} \cdot \boldsymbol{J}
\end{gathered}
$$

- The couple is then given by

$$
\boldsymbol{G}=\dot{\boldsymbol{J}}=\boldsymbol{\omega} \times \boldsymbol{J}
$$

- Note that $\boldsymbol{I}$ must be specified with its origin and with its set of axes.


## Properties of $I$

- $\boldsymbol{I}$ is a symmetric tensor. It therefore has three real eigenvalues and three perpendicular eigenvectors.
- With respect to the eigenvector basis:

$$
\begin{gathered}
\boldsymbol{I}^{\prime}=\left(\begin{array}{ccc}
I_{1} & \cdot & \cdot \\
\cdot & I_{2} & \cdot \\
\cdot & \cdot & I_{3}
\end{array}\right) \\
\boldsymbol{J}_{\alpha}=I_{a} \omega_{\alpha} \\
{\left[\begin{array}{ll}
\text { No sum }]
\end{array}\right.} \\
T=\frac{1}{2} I_{\alpha} \omega_{\alpha}^{2} \\
{[\text { Sum }]}
\end{gathered}
$$

- The eigenvector axes are called the principal axes, and the Is are called the principal moments of inertia.
- An alternative way to think of this is that the principal axes are ones around which objects are "happy" to rotate without any torque being applied.
- In $\omega$-space, surfaces of constant $T$ form an ellipsoid, with axes of length $\propto I_{\alpha}^{-1 / 2}$. Also, in $\omega$-space:

$$
\operatorname{grad} T=I_{\alpha} \omega_{\alpha}=\boldsymbol{J}
$$

So $\boldsymbol{J}$ is perpendicular to surfaces of constant $\boldsymbol{T}$ at $\boldsymbol{\omega}$.

- We can classify the principal axes as follows:
o Spherical tops - all the $I$ are equal, and $\boldsymbol{J}=I \boldsymbol{\omega}$, with $I$ scalar. The body is isotropic with the same $I$ about any axis (eg: sphere, cube).
o Symmetrical tops - $I_{1}=I_{2} \neq I_{3} . \mathbf{e}_{3}$ axis is unique, but $\mathbf{e}_{1}$ and $\mathbf{e}_{2}$ are any two mutually perpendicular vectors perpendicular to $\mathbf{e}_{3}$ (eg: lens, cigar).
o Asymmetrical tops - all Is different, and axes are unique.
- Consider any two Is:

$$
I_{1}+I_{2}=\sum m\left(y^{2}+z^{2}+x^{2}+z^{2}\right)=I_{3}+2 \sum m z^{2} \geq I_{3}
$$

So no $I$ can be larger than the sum of the other two. Furthermore, if $\boldsymbol{z}=\boldsymbol{O}$ for every particle (ie: if we have a lamina), then

$$
I_{3}=I_{1}+I_{2}
$$

- Consider an axis at a distance a away from a principal axis and parallel to it, and let $\mathbf{r}$ be the distance of each particle from the principal axis. Then:

$$
I=\sum m(\boldsymbol{r}+\boldsymbol{a}) \cdot(\boldsymbol{r}+\boldsymbol{a})=I_{0}+M a^{2}+2 \underbrace{\left(\sum m \boldsymbol{r}\right)}_{\substack{=0 \\ \text { = when } r \text { measured } \\ \text { relive to C of } \mathrm{M}}} \cdot \boldsymbol{a}=I_{0}+M a^{2}
$$

This is the Parallel Axis Theorem, where each vector is considered to be a projection in a plane perpendicular to the axes.

## Two Basic Problems

- You whack it - what happens? Steps for solution:
o Define principal axes with a sensible origin.
o Calculate an expression for $\boldsymbol{J}$ in terms of the impulse:

$$
\boldsymbol{J}=\int \boldsymbol{\tau} \mathrm{d} t=\int \boldsymbol{r}_{B} \times \boldsymbol{F} \mathrm{d} t=\boldsymbol{r}_{B} \times \int \boldsymbol{F} \mathrm{d} t=\boldsymbol{r}_{B} \times \boldsymbol{P}
$$

Where $B$ is the point at which the whack occurred, and $\boldsymbol{r}_{B}$ can be taken out of the integral because the whack is assumed to be instantaneous.
o Work out an expression for $\boldsymbol{J}$ in terms of $\boldsymbol{\omega}$, using the moments of inertia.
o Equate the two expressions for $\boldsymbol{J}$.
o Work out the motion of the CM using standard linear mechanics.
o Note: The obvious origin to use is the CM, but other origins can be used subject to the provisos above for using $\boldsymbol{\tau}=\dot{\boldsymbol{J}}$. So a pivot, for example, is fine to use.

- You apply a torque - what's the frequency of rotation?
o Define principal axes with a sensible origin (eg: the CM - see above).
o Find an expression for $\boldsymbol{\omega}$ in these axes (with unknown magnitude), and find a corresponding expression for $\boldsymbol{J}$, using the principal moments of inertia.
o Find $\mathrm{d} \boldsymbol{L} / \mathrm{d} t=\boldsymbol{\omega} \times \boldsymbol{L}$.
o Calculate the torque $(=\boldsymbol{r} \times \boldsymbol{F})$ and equate it with $\mathrm{d} \boldsymbol{L} / \mathrm{d} t$.


## Free Motion - Euler's Equation

- Free precession is a situation in which $\boldsymbol{F}=\boldsymbol{0}$ and $\boldsymbol{G}=\boldsymbol{0}$. In such a case, $\boldsymbol{J}$ is constant. $\boldsymbol{\omega}$ is constant if $\boldsymbol{J}$ is along one of the principal axes, but otherwise, it will change direction, and perhaps even magnitude.
- We use the Euler Equations to analyse this problem.
- The rate of change of angular momentum vector in the principal-axes frame (which is rotating around with the body) and the lab frame are related by

$$
\left[\frac{\mathrm{d} \boldsymbol{J}}{\mathrm{~d} t}\right]_{\mathrm{lab}}=\left[\frac{\mathrm{d} \boldsymbol{J}}{\mathrm{~d} t}\right]_{\mathrm{PA}}+\boldsymbol{\omega} \times \boldsymbol{J}
$$

- Now, let's assume that a couple $\boldsymbol{G}$ is being applied in the lab frame. We know that

$$
\boldsymbol{G}=\left[\frac{\mathrm{d} \boldsymbol{J}}{\mathrm{~d} t}\right]_{\mathrm{lab}}
$$

Therefore, using the equations above:

$$
\boldsymbol{G}=\left[\frac{\mathrm{d} \boldsymbol{J}}{\mathrm{~d} t}\right]_{\mathrm{PA}}+\boldsymbol{\omega} \times \boldsymbol{J}
$$

- Finally, we note that in the principal axes frame, $\boldsymbol{J}=\left(I_{1} \omega_{1}, I_{2} \omega_{2}, I_{3} \omega_{3}\right)$. Therefore, casting both sides of this equation into the principal axes frame only

$$
\tau_{1}=I_{1} \dot{\omega}_{1}+\left(I_{3}-I_{2}\right) \omega_{3} \omega_{2}
$$

And similarly with any cyclic permutation of indices.

- A few notes:
o All the quantities in this equation are measured with respect to the body frame (which is moving). This is the advantage of these equations - all we have to consider is the forces that the body "feels".
o The two terms of the RHS refer to two types of ways $\boldsymbol{J}$ can change - because it can change in the body frame and also because the body frame is itself rotating.


## Free Motion - Examples

- Free Symmetric Top
o For a symmetrical top ( $I_{1}=I_{2}=I$ ) which is free in space (ie: no torque) the Euler Equations become

$$
\begin{gathered}
I \dot{\omega}_{1}+\left(I_{3}-I\right) \omega_{3} \omega_{2}=0 \\
I \dot{\omega}_{2}+\left(I-I_{3}\right) \omega_{1} \omega_{3}=0 \\
I_{3} \dot{\omega}_{3}=0
\end{gathered}
$$

o The last equation implies $\omega_{3}$ is constant. Let's define

$$
\Omega=\frac{I_{3}-I}{I} \omega_{3}
$$

Then the general solution of the first two equations becomes:

$$
\binom{\omega_{1}}{\omega_{2}}=A\binom{\cos [\Omega t+\phi]}{\cos [\Omega t+\phi]}
$$

## o Interpretation from the body frame

- In the body frame, $\omega_{1}$ and $\omega_{2}$ seem to form a circle in the $x-y$ plane, with frequency $\Omega$. How high that circle is depends on $\boldsymbol{\omega}_{3}$.
- $\boldsymbol{L}$ could be above $\boldsymbol{\omega}$ (if $I_{3}>I$ - an oblate top) or below $\boldsymbol{\omega}$ (if $I_{3}<I$-a prolate top).


## o Interpretation from the fixed lab frame

- In that case, the Euler Equations are useless, because they deal with the body frame, so we express things from scratch, but in terms of the body frames:

$$
\begin{gathered}
\boldsymbol{\omega}=\left(\omega_{1} \hat{\boldsymbol{x}}_{1}+\omega_{2} \hat{\boldsymbol{x}}_{2}\right)+\omega_{3} \hat{\boldsymbol{x}}_{3} \\
\boldsymbol{L}=I\left(\omega_{1} \hat{\boldsymbol{x}}_{1}+\omega_{2} \hat{\boldsymbol{x}}_{2}\right)+I_{3} \omega_{3} \hat{\boldsymbol{x}}_{3}
\end{gathered}
$$

$$
\Downarrow
$$

$$
\omega=\frac{L}{I}-\Omega \hat{\boldsymbol{x}}_{3}
$$

With $\Omega$ defined as above.

- This linear relationship between $\boldsymbol{\omega}, \boldsymbol{L}$ and $\hat{\boldsymbol{x}}_{3}$ implies that they are in the same plane.
- Furthermore, the rate of change of $\hat{\boldsymbol{x}}_{3}$ is $\boldsymbol{\omega} \times \hat{\boldsymbol{x}}_{3}$, because it only changes as a result of the rotation. So

$$
\frac{\mathrm{d} \hat{\boldsymbol{x}}_{3}}{\mathrm{~d} t}=\left(\frac{\boldsymbol{L}}{I}-\Omega \hat{\boldsymbol{x}}_{3}\right) \times \hat{\boldsymbol{x}}_{3}=\left(\frac{\boldsymbol{L}}{I}\right) \times \hat{\boldsymbol{x}}_{3}
$$

This is equivalent to $\hat{\boldsymbol{x}}_{3}$ rotating at a frequency $L / I$.

- It turns out that we can interpret $\boldsymbol{\omega}$ as follows

$$
\boldsymbol{\omega}=\overbrace{\frac{\boldsymbol{L}}{I}}^{\text {Motion of body around } L}-\overbrace{\Omega \hat{\boldsymbol{x}}_{3}}^{\begin{array}{c}
\text { Motion of body about } \\
\text { its own axis }
\end{array}}
$$

## - Heavy Symmetric Top

o Here, we must define the Euler angles as follows

o The total angular velocity is then given by

$$
\boldsymbol{\omega}=\overbrace{\dot{\psi} \hat{\boldsymbol{x}}_{3}}^{\text {Rotation of top }}+\overbrace{\dot{\theta} \hat{\boldsymbol{x}}_{1}+\dot{\phi} \boldsymbol{z}}^{\text {Motion of top itself }}
$$

Which can be expressed in terms of the body-frames only:

$$
\begin{aligned}
\boldsymbol{\omega} & =\dot{\psi} \hat{\boldsymbol{x}}_{3}+\dot{\theta} \hat{\boldsymbol{x}}_{1}+\dot{\phi}\left(\hat{\boldsymbol{x}}_{3} \cos \theta+\hat{\boldsymbol{x}}_{2} \sin \theta\right) \\
\boldsymbol{\omega} & =(\dot{\psi}+\dot{\phi} \cos \theta) \hat{\boldsymbol{x}}_{3}+(\dot{\phi} \sin \theta) \hat{\boldsymbol{x}}_{2}+\dot{\theta} \hat{\boldsymbol{x}}_{1}
\end{aligned}
$$

0

## Dynamics - Normal Modes

## Introduction

- A normal mode of a system is an oscillation that has a single frequency.
- All the more general oscillations of the system can be expressed as superpositions of these normal modes.


## General approach

- Consider a system defined by generalised coordinates $q_{i}$ and acted on by forces $F_{i}$, moving in a potential well $U(\mathrm{x})$, and moving elastically.
- The kinetic energy, $T$, is then given by

$$
T=\frac{1}{2} \sum \sum m_{i}\left|\dot{\boldsymbol{\xi}}_{j}\left(q_{i}\right)\right|^{2}
$$

Where $\sum_{i} \boldsymbol{\xi}_{j}\left(q_{i}\right)$ is the Cartesian coordinate of the $j^{\text {th }}$ part of the system, taken about an equilibrium, where all the $\boldsymbol{\xi}_{j}$ are 0 . Expanding about that equilibrium:

$$
\begin{gathered}
\sum_{i} \boldsymbol{\xi}_{j}\left(q_{i}\right)=\sum_{i} \boldsymbol{\xi}_{j}\left(q_{i, \text { eq }}\right)+\left.\frac{\partial \boldsymbol{\xi}_{j}}{\partial q_{i}}\right|_{\mathrm{eq}} q_{i}+\cdots \\
\left.\sum_{i} \dot{\boldsymbol{\xi}}_{j}\left(q_{i}\right) \approx \sum_{i} \frac{\partial \boldsymbol{\xi}_{j}}{\partial q_{i}}\right|_{\mathrm{eq}}
\end{gathered}
$$

And so:

$$
T=\frac{1}{2} \sum \sum M_{i j} \dot{q}_{i} \dot{q}_{j}=\frac{1}{2} \dot{\boldsymbol{q}}^{T} \boldsymbol{M} \dot{\boldsymbol{q}}
$$

Where

$$
M_{i j}=\left.\left.\sum \sum m \frac{\partial \boldsymbol{r}}{\partial q_{i}}\right|_{\mathrm{eq}} \frac{\partial \boldsymbol{r}}{\partial q_{j}}\right|_{\mathrm{eq}}
$$

- Consider the potential energy, about a point of equilibrium (ie: a minimum in $\boldsymbol{U}$ ) at which all the $\boldsymbol{q}_{i}$ are chosen to be 0 .

$$
\begin{gathered}
U(x)=U_{0}+\left.\underbrace{\sum \frac{\partial U}{\partial q_{i}}}_{0 \text { since at a minimum }}\right|_{\mathrm{Eq}} q_{i}+\left.\sum \sum \frac{1}{2} \frac{\mathrm{~d}^{2} U}{\mathrm{~d} x_{j} \mathrm{~d} x_{i}}\right|_{x_{0}} q_{i} q_{j}+\cdots \\
U(x)=U_{0}+\frac{1}{2} \sum \sum K_{i j} q_{i} q_{j}+\cdots \\
U(x)=U_{0}+\frac{1}{2} \boldsymbol{q}^{T} \boldsymbol{K} \boldsymbol{q}
\end{gathered}
$$

- The total energy is then

$$
\begin{gathered}
E=U_{0}+\frac{1}{2} \sum \sum M_{i j} \dot{q}_{i} \dot{q}_{j}+\frac{1}{2} \sum \sum K_{i j} q_{i} q_{j} \\
\frac{\mathrm{~d} E}{\mathrm{~d} t}=\frac{1}{2} \sum \sum 2 \dot{q}_{i}\left(M_{i j} \ddot{q}_{j}+K_{i j} q_{j}\right)=0 \\
\frac{\mathrm{~d} E}{\mathrm{~d} t}=\sum \sum \dot{q}_{i}\left(M_{i j} \ddot{q}_{j}+K_{i j} q_{j}\right)=0
\end{gathered}
$$

- [Non rigorous argument] - the equations of motion are then:

$$
\begin{gathered}
\sum \sum M_{i j} \ddot{q}_{j}+\sum \sum K_{i j} q_{j}=0 \\
\boldsymbol{M} \ddot{\boldsymbol{q}}+\boldsymbol{K} \boldsymbol{q}=0
\end{gathered}
$$

- If we seek normal modes of the form $\boldsymbol{q}(t)=\boldsymbol{Q} e^{i \omega t}$, we get:

$$
\left(\boldsymbol{K}-\omega^{2} \boldsymbol{M}\right) \boldsymbol{Q}=0
$$

Non-trivial solutions only exist if

$$
\operatorname{det}\left(\boldsymbol{K}-\omega^{2} \boldsymbol{M}\right)=0
$$

This defines the $\omega^{2}$ normal mode frequencies.

- In practice, the steps are:
o Find the $\boldsymbol{K}$ and $\boldsymbol{M}$ matrices by writing them out in terms of the variables of the system, and comparing with

$$
T=\frac{1}{2} \dot{\boldsymbol{q}}^{T} \boldsymbol{M} \dot{\boldsymbol{q}} \quad U=U_{0}+\frac{1}{2} \boldsymbol{q}^{T} \boldsymbol{K} \boldsymbol{q}
$$

Both matrices must be symmetric.
o Use the determinant method above.

## Dynamics - Elasticity

## Introduction

- Hooke's Law states that

$$
\frac{\stackrel{F}{A}}{\stackrel{\text { Stress }}{ }}=E \frac{\stackrel{\text { Strain }}{\Delta l}_{l}^{\Delta}}{}
$$

Where
o $\boldsymbol{F}$ is the force applied to a block of material over an area $\boldsymbol{A}$.
o $\Delta l$ is the extension of the block in the direction of $\boldsymbol{F}$.
o $l$ is the original, relaxed length of the block in that direction.
o $\boldsymbol{E}$ is the Young's Modulus of the material.

- Furthermore, it states that

$$
\frac{\Delta w}{w}=-\sigma \frac{\Delta l}{l}
$$

Where $\boldsymbol{\Delta} \boldsymbol{w}$ is the length of the block in any direction perpendicular to that of $l$.

- For an isotropic material, $\boldsymbol{E}$ and $\boldsymbol{\sigma}$ are all we need to define the elastic properties of the material.
- Since these equations are all linear, the principle of superposition applies. If we have several forces, the displacements will be the sum of the displacements with the forces acting individually.


## Uniform Strain - the Bulk Modulus

- Consider a rectangular block in a pressure tank, say, with identical stress $\boldsymbol{p}$ on every face.
- Consider one direction - the change in length $\Delta l$ in that direction is given by

$$
\begin{gathered}
\frac{\Delta l}{l}=\overbrace{-\frac{p}{E}}^{\substack{\text { Due to pressure } \\
\text { in that direction }}}+\overbrace{\sigma \frac{p}{E}+\sigma \frac{p}{E}}^{\begin{array}{c}
\text { Due to pressure in } \\
\text { oter directions }
\end{array}} \\
\frac{\Delta l}{l}=-\frac{1-2 \sigma}{E} p
\end{gathered}
$$

The problem is symmetrical, so the value will be the same for all directions.

- Now, consider the change in volume

$$
\frac{\Delta V}{V}=\frac{\Delta x}{x}+\frac{\Delta y}{y}+\frac{\Delta z}{z}
$$

We therefore have

$$
\frac{\Delta V}{V}=-3 \frac{1-2 \sigma}{E} p
$$

- We can then define the bulk modulus

$$
K=\frac{E}{3(1-2 \sigma)}
$$

Such that the change of volume as a result of the stress $\boldsymbol{p}$ is

$$
p=-K \frac{\Delta V}{V}
$$

## Shear Strain - the Shear Modulus

- Consider a cube with face area $\boldsymbol{A}$ and with shear forces acting on it


If cut the cube along the diagonals $\boldsymbol{A}$ and $\boldsymbol{B}$, we find that
o There is a stretch normal to $A$, of magnitude $\boldsymbol{F} \sqrt{2}$.
o There is a compression normal to $B$, of magnitude $\boldsymbol{F} \sqrt{2}$.
And each of these diagonal faces has area $A \sqrt{2}$.

- The lengthening of the diagonal $d$ will therefore be equal to the lengthening of $d$ in the following case:


From above, this is given by:

$$
\begin{gathered}
\frac{\Delta d}{d}=\frac{1}{E} \frac{F \sqrt{2}}{A \sqrt{2}}+\sigma \frac{1}{E} \frac{F \sqrt{2}}{A \sqrt{2}} \\
\frac{\Delta d}{d}=\frac{1+\sigma}{E} \frac{F}{A}
\end{gathered}
$$

By symmetry, the other diagonal is shortened by the same amount.

- It is often useful to have this as a function of the twist angle:


From this diagram, it is (reasonably) clear that

$$
\delta=\Delta d \sqrt{2} \quad d=\ell \sqrt{2}
$$

Therefore

$$
\theta \approx \frac{\delta}{l}=\frac{\Delta d \sqrt{2}}{l}=2 \frac{\Delta d}{d}=\frac{2(1+\sigma)}{E} \frac{F}{A}
$$

- We therefore define the shear modulus as

$$
\mu=\frac{E}{2(1+\sigma)}
$$

Such that

$$
g=\mu \theta
$$

Where $g$ is the shear stress $=\boldsymbol{F} / \boldsymbol{A}$.

## Formal Definitions

- Stress
o Defined in terms of force/unit area transmitted across planes in the medium.
o Requires a tensor. We define

o We can then show that the force on any arbitrary area element is

$$
\boldsymbol{F}=\boldsymbol{\tau} \mathrm{d} \boldsymbol{S}
$$

o The tensor must be symmetric - consider a small cube side $\mathrm{d} x$. Because the cube must be in equilibrium, the forces on it are as follows:


The net couple on the cube is

$$
\left(S_{x y}-S_{y x}\right) \mathrm{d} x
$$

But there must be no torque on the cube, or it'd spin! So

$$
S_{x y}=S_{y x}
$$

o The stress tensor is diagonal for suitable choices of axes.
o The stress in a solid material is therefore described by a tensor field.

- Strain
o When a material is put under strain, a point $(x, y, z)$ in it is moved to a point $(x+X, y+Y, z+Z)$.
o The derivatives of these $X, Y$ and $Z$ contain information about the strain.
o As we saw before, it's worth considering two kinds of strain
- For the normal strains, we define:

$$
e_{x x}=\frac{\partial X}{\partial x} \quad e_{y y}=\frac{\partial Y}{\partial y} \quad e_{z z}=\frac{\partial Z}{\partial z}
$$

For example, if we consider stress perpendicular to the $\boldsymbol{x}$ direction in a cube initially of side $\Delta x$, it'll increase by $e_{x x} \Delta x$ :


- Now, for the shear stresses, consider

[The expression for the angles are tricky to see - but consider that $X$ is the change in $x \ldots$...] We then simply define

$$
e_{x y}=e_{y x}=\frac{1}{2}\left(\frac{\partial Y}{\partial x}+\frac{\partial X}{\partial y}\right)
$$

This ensures that if the block simply rotates (ie: $\partial Y / \partial y=\partial X / \partial x)$, these strains are 0.
o So in general, we define


So, for example

$$
X=e_{x x} x+e_{x y} y+e_{x z} z
$$

o The tensor is also symmetric, due to the $e_{x y}=e_{y x}$ condition.
o If the strains are non-homogenous, we sit down and cry.

- The relation between them
o Each component of $e$ is related to each component of $\tau$ - this gives, overall, a fourth-rank tensor of elasticity relating the two:

$$
\tau_{i j}=C_{i j k l} e_{k l}
$$

(Using the summation convention).
o It looks like there are $9^{2}=81$ coefficients in $C$, and that $\mathbf{8 1}$ numbers are therefore required to define the elastic properties of a material! However, we note that since $S$ and $e$ are symmetry, we must be able to swap $i j$ and $k l$ in $C$ without changing a thing, so there can be at most 36 different coefficients.
o If the material is isotropic, though, $C$ must be completely frameindependent. As such, we must be able to express it in terms of the tensor $\delta_{i j}$. There are only two ways of doing this that are also invariant under $i \leftrightarrow j$ and $l \leftrightarrow k$, and so

$$
C_{i j k l}=\lambda\left(\delta_{i j} \delta_{k l}\right)+\mu\left(\delta_{i k} \delta_{j l}+\delta_{i l} \delta_{j k}\right)
$$

So an isotropic material only requires two constants ( $E$ and $\sigma$, for example). And we have

$$
S_{i j}=\lambda e_{k k} \delta_{i j}+2 \mu e_{i j}
$$

## Examples - Statics

- Thin tube in torsion
o Consider a thin tube being twisted an angle $\phi$

o We first note that

$$
\theta=\frac{r \phi}{l}
$$

o Next, consider a small square (dotted above) and its deformation as a result of the twist:


From the previous result:

$$
\begin{gathered}
\frac{F}{\ell \Delta r}=\mu \theta \\
F=\mu \frac{r \phi}{L} \ell \Delta r
\end{gathered}
$$

o This force contributes a torque $\boldsymbol{\Delta} \boldsymbol{\tau}$ to the rod

$$
\Delta \tau=r F=\mu \frac{r^{2} \phi}{L} \ell \Delta r
$$

o Considering these bits around the whole rod, so that $\ell \rightarrow 2 \pi r$, we get

$$
\tau=2 \pi \mu \frac{r^{3} \Delta r}{L} \phi
$$

- Wire in torsion

For a wire, we simply integrate the above from $r=0$ to the total radius, giving

$$
\tau=\mu \frac{\pi r^{4}}{2 L} \phi
$$

- Can under pressure
o Consider a can of thickness $t$ with closed ends with an internal pressure $p$.
o Let the tangential stress in the walls be $\tau_{\theta}$, and consider half the can


The forces ( $=$ stress $\times$ area) must balance, so

$$
\begin{gathered}
\tau_{\theta} \times 2 t=p \times 2 r \\
\tau_{\theta}=\frac{p r}{t}
\end{gathered}
$$

o Let the axial stress in the walls be $\sigma_{z}$, and consider one of the ends. By the same logic as above

$$
\begin{gathered}
\tau_{z} \times 2 \pi r t=p \times \pi r^{2} \\
\tau_{z}=\frac{p r}{2 t}
\end{gathered}
$$

- Bent beam
o Consider a beam of length $\boldsymbol{L}$, held in a bent position.
o We only consider longitudinal strains (valid for small deflections and thin beams).
o Clearly, the bits at the top of the beam will be stretched, while those at the bottom will be compressed. Somewhere in between, there'll be a neutral surface - neither stretched nor compressed.
o Consider a small segment length $\ell$ of the bent beam:

o The amount of stretching and compression at any point is proportional to the distance from the neutral surface, $\boldsymbol{y}$. The constant of proportionality is $\ell / R$. As such

$$
\frac{\Delta \ell}{\ell}=\operatorname{Strain}=\frac{y}{R}
$$

o Clearly, there'll be forces to the left above the neutral surface, and vice versa. We therefore have

$$
\begin{aligned}
& \frac{\Delta F}{\Delta A}=E \frac{\Delta \ell}{\ell} \\
& \Delta F=\frac{E}{R} y \Delta A
\end{aligned}
$$

o The total torque produced about the neutral line is given by

$$
\begin{aligned}
\boldsymbol{\tau} & =\int_{\text {secrion }} y \mathrm{~d} F \\
= & \frac{E}{R} \int_{\text {section }} y^{2} \mathrm{~d} A \\
& B=\frac{E I}{R}
\end{aligned}
$$

o Now, consider a beam loaded with weights given by $\boldsymbol{W}(x)$, where $\boldsymbol{W}$ is the force per unit length. Consider the statics of a small segment of the beam:


Notes:

- Due to the bending moment, some vertical forces are produced. Ignoring products of infinitesimal quantities, we can write, at that point

$$
\begin{aligned}
S \mathrm{~d} x & =\mathrm{d} B \\
B=\frac{E I}{R} & =\int S \mathrm{~d} x
\end{aligned}
$$

[Effectively, we're saying that due to the $\mathrm{d} S$ needed to balance $\mathrm{d} F$, the bending moment must change]

- The downwards loading force needs to be balanced by a difference in the upwards stress

$$
\begin{gathered}
\mathrm{d} S=\mathrm{d} F=W \mathrm{~d} x \\
S^{\prime}=W
\end{gathered}
$$

o Now, for small deflections

$$
y^{\prime \prime}=1 / R
$$

o As such, we can conclude

$$
E I y^{\prime \prime \prime \prime}=W(x)
$$

o Boundary conditions for various cases are as follows

- At a free end, $\boldsymbol{S}$ and $\boldsymbol{B}$ are clearly 0 , and so $y^{\prime \prime}=y^{\prime \prime \prime}=0$.
- At a cantilevered end, $y$ and $y^{\prime}$ are given (usually 0 ).
o Finding $y$ is then simply a question of solving that differential equation. However, there are a few tricky points
- All forces must be considered when writing down $\boldsymbol{W}(x)$, including reactions at contacts. Most often, $\boldsymbol{W}$ will be a series of $\delta$-functions.
- Sign conventions:
- Downwards $W \rightarrow$ positive.
- The resulting $y$ obtained is downwards $\rightarrow$ positive, because the way the radius of curvature is specified.
- However, be very careful - sometimes, the convention appears to be reversed because the bar curves downwards, and so $-1 / R=y^{\prime \prime}$.
- Don't worry too much about boundary conditions for $y^{\prime \prime \prime}$ - just integrate $\delta$-functions from 0 to $L$ (for a free end, this is fully justified). Remember that there'll often be a $\delta$-functions at the very end of the range, which might help satisfy the boundary conditions.
- From then on, boundaries are just provided. Just also remember to make the $y^{\prime \prime}, y^{\prime}$ and $y$ continuous.
- The couple provided by a cantilever can simply be worked out by evaluating $\boldsymbol{B}=\boldsymbol{E I} \boldsymbol{y}^{\prime \prime}$ at that point.
- It is sometimes easier to simply write down $y^{\prime \prime}$, the bending moment from physical considerations.
o The Euler Strut is a beam buckled between two walls:


If we take $y$ upwards, then the bending moment on any point is

$$
\begin{gathered}
B=-F y \\
y^{\prime \prime}=-\frac{F}{E I} y \\
y=A \sin \left[x \sqrt{\frac{F}{E I}}\right]
\end{gathered}
$$

Applying the boundary condition that $y=0$ at $x=L$ :

$$
F=\frac{\pi^{2} E I}{L^{2}}
$$

This is independent of displacement (but only while $y^{\prime \prime}=1 / R$ holds).
o The Reciprocity Theorem states that

> "The deflection at $Q$ due to a load at $P$ is the same as the deflection at $P$ due to the same
> load at $Q$ "

To prove, say $P_{P Q}$ means "the deflection at $P$ due to the load at $Q "$. Consider loading first $P$ and then $Q$. The energy stored is

$$
E=F\left[\frac{P_{P P}}{2}+\frac{P_{Q Q}}{2}+P_{P Q}\right]
$$

The same result must be applied the other way round, so

$$
P_{P Q}=P_{Q P}
$$

## Dynamics of Rigid Bodies

- Consider a small volume $\boldsymbol{V}$ of the material. It will have both external forces acting on it (eg: gravity) and internal forces (eg: elastic stresses).

$$
\boldsymbol{F}_{\text {ext }}+\boldsymbol{F}_{\text {int }}=\int \rho \ddot{\boldsymbol{r}} \mathrm{d} V
$$

- Every small particle in the volume experiences the external force, though, so $\boldsymbol{F}_{\text {ext }}$ is given by a volume integral.

$$
\begin{gathered}
\boldsymbol{F}_{\text {int }}=\int \overbrace{\left(-\boldsymbol{f}_{\text {ext }}+\rho \ddot{\boldsymbol{r}}\right)}^{\text {Define this }=f} \mathrm{~d} V \\
\boldsymbol{F}_{\text {int }}=\int_{V} \boldsymbol{f} \mathrm{~d} V
\end{gathered}
$$

On the other hand, only the particles at the edge of the volume experience the elastic force from surrounding media, and so $\boldsymbol{F}_{\text {int }}$ is given by an area integral

$$
\int_{\mathrm{A}} \boldsymbol{f}_{\text {int }} \mathrm{d} A=\int_{V} \boldsymbol{f} \mathrm{~d} V
$$

- We have, however, defined that the force in the $x$-direction, say, is

$$
\mathrm{d} F_{x}=\left(S_{x x} \boldsymbol{i}+S_{x y} \boldsymbol{j}+S_{x z} \boldsymbol{k}\right) \cdot \mathrm{d} \boldsymbol{A}
$$

And so, taking only the $x$ component of the integral above

$$
\int_{\mathrm{A}}\left(S_{x x} \boldsymbol{i}+S_{x y} \boldsymbol{j}+S_{x z} \boldsymbol{k}\right) \cdot \mathrm{d} \boldsymbol{A}=\int_{V} f_{x} \mathrm{~d} V
$$

- Using the Divergence Theorem on the LHS

$$
\int_{V}\left(\frac{\partial S_{x x}}{\partial x}+\frac{\partial S_{x y}}{\partial y}+\frac{\partial S_{x z}}{\partial z}\right) \mathrm{d} V=\int_{V} f_{x} \mathrm{~d} V
$$

Removing the volume integrals (because this is true for any volume):

$$
\boldsymbol{f}_{i}=\partial S_{i j} / \partial x_{j}
$$

(Using the summation convention).

- Now, using $\tau_{i j}=\lambda e_{k k} \delta_{i j}+2 \mu e_{i j}$ (isotropic material), we obtain

$$
\boldsymbol{f}=(\lambda+\mu) \nabla(\nabla \cdot \boldsymbol{u})+\mu \nabla^{2} \boldsymbol{u}
$$

Where $\boldsymbol{u}$ is the internal displacement in the solid.

