Dynamics – Rigid Body Dynamics

Introduction

- A rigid body is a many-particle system in which the distance between particles is fixed. The location of all particles is described by 6 coordinates – 3 spatial and 3 angular.
- The velocity is determined by v, the velocity of the CoM and ω , the angular velocity.
- The basic two equations of angular motion are

$$M\ddot{R} = F_0$$

The centre of mass moves as if it were a single particle under the action of a force F_{o} .

$$\dot{m{J}}=m{G}_{_0}$$

The rate of change of angular momentum is equal to the total applied couple.

- Other basic equations:
 - The velocity v of a particle at a distance r from an axis around which a rotation at speed ω is happening is

$$oldsymbol{v} = oldsymbol{\omega} imes oldsymbol{r}$$

• For similar reasons:

$$\frac{\mathrm{d}\boldsymbol{J}}{\mathrm{d}t} = \boldsymbol{\omega} \times \boldsymbol{J}$$

• Angular speeds are **additive**. To if *frame 1* is rotating with $\omega_{1 \text{ wrt } 2}$ with respect to *frame 2*, which is rotating with $\omega_{2 \text{ wrt } 3}$ with respect o *frame 3*, then

$$\boldsymbol{\omega}_{1\text{ wrt }3} = \boldsymbol{\omega}_{1\text{ wrt }2} + \boldsymbol{\omega}_{2\text{ wrt }3}$$

Relating J and ω

• If the body is rotating at $\boldsymbol{\omega}$, the **total angular momentum** is given by

$$\begin{aligned} \boldsymbol{J} &= \sum \boldsymbol{r} \times \boldsymbol{p} \\ &= \sum \boldsymbol{r} \times \boldsymbol{m} (\boldsymbol{\omega} \times \boldsymbol{r}) \\ &= \sum \boldsymbol{m} \big[r^2 \boldsymbol{\omega} - (\boldsymbol{\omega} \cdot \boldsymbol{r}) \boldsymbol{r} \big] \\ &= \sum \boldsymbol{m} \big[r^2 \boldsymbol{\omega} - (\omega_x x + \omega_y y + \omega_z z) \boldsymbol{r} \big] \end{aligned}$$

In detail

$$\boldsymbol{J} = \underbrace{ \begin{pmatrix} \sum m(y^2 + z^2) & -\sum mxy & -\sum mxz \\ -\sum mxy & \sum m(x^2 + z^2) & -\sum myz \\ -\sum mxz & -\sum myz & \sum m(x^2 + y^2) \end{pmatrix}}_{\boldsymbol{I}} \boldsymbol{\omega}$$

[The non-diagonal elements are fairly easy to derive. The diagonal ones should actually have $x^2 + y^2 + z^2$, because one of the terms is always knocked out by the second term in the sum]. In other words, J is proportional to ω , but not necessarily parallel to it.

- The off-axes elements are rather hard to understand they correspond to the fact that looking at a particle at a given instant, it's impossible to tell exactly around which axis it's moving.
- Also, we can find the **kinetic energy**

$$T = \sum_{n=1}^{\infty} \frac{1}{2} m [(\boldsymbol{\omega} \times \boldsymbol{r}) \cdot (\boldsymbol{\omega} \times \boldsymbol{r})]$$
$$= \sum_{n=1}^{\infty} \frac{1}{2} m [\boldsymbol{\omega} \cdot \boldsymbol{r} \times (\boldsymbol{\omega} \times \boldsymbol{r})]$$
$$T = \frac{1}{2} \boldsymbol{\omega} \cdot \boldsymbol{J}$$

• The couple is then given by

$$oldsymbol{G}=\dot{oldsymbol{J}}=oldsymbol{\omega} imesoldsymbol{J}$$

• Note that *I* must be specified with its origin and with its set of axes.

Properties of I

- *I* is a symmetric tensor. It therefore has three real eigenvalues and three perpendicular eigenvectors.
- With respect to the **eigenvector basis**:

$$\boldsymbol{I}' = \begin{pmatrix} I_1 & \cdot & \cdot \\ \cdot & I_2 & \cdot \\ \cdot & \cdot & I_3 \end{pmatrix}$$
$$\boldsymbol{J}_{\alpha} = I_{\alpha} \omega_{\alpha} \quad [\text{No sum}]$$
$$\boldsymbol{T} = \frac{1}{2} I_{\alpha} \omega_{\alpha}^2 \quad [\text{Sum}]$$

- The eigenvector axes are called the principal axes, and the *I*s are called the principal moments of inertia.
- An alternative way to think of this is that the principal axes are ones around which objects are "happy" to rotate without any torque being applied.
- In ω -space, surfaces of constant T form an **ellipsoid**, with **axes** of length $\propto I_{\alpha}^{-1/2}$. Also, in ω -space:

grad
$$T = I_{\alpha}\omega_{\alpha} = \boldsymbol{J}$$

So J is perpendicular to surfaces of constant T at ω .

- We can classify the **principal axes** as follows:
 - Spherical tops all the *I* are equal, and *J* = *I\omega*, with *I* scalar. The body is isotropic with the same *I* about any axis (eg: sphere, cube).
 - Symmetrical tops I₁ = I₂ ≠ I₃. e₃ axis is unique, but e₁ and e₂ are any two mutually perpendicular vectors perpendicular to e₃ (eg: lens, cigar).
 - Asymmetrical tops all *I*s different, and axes are unique.
- Consider any two *I*s:

$$I_1 + I_2 = \sum m(y^2 + z^2 + x^2 + z^2) = I_3 + 2\sum mz^2 \ge I_3$$

So no *I* can be larger than the sum of the other two. Furthermore, if z = 0 for every particle (ie: if we have a lamina), then

$$I_3 = I_1 + I_2$$

 Consider an axis at a distance a away from a principal axis and parallel to it, and let r be the distance of each particle from the principal axis. Then:

$$I = \sum m(\boldsymbol{r} + \boldsymbol{a}) \cdot (\boldsymbol{r} + \boldsymbol{a}) = I_0 + Ma^2 + 2 \underbrace{\left(\sum_{\substack{n=0 \text{ when } \boldsymbol{r} \text{ measured} \\ \text{relative to C of M}}} \cdot \boldsymbol{a} = I_0 + Ma^2$$

This is the **Parallel Axis Theorem**, where each vector is considered to be a **projection** in a plane **perpendicular** to the **axes**.

Two Basic Problems

- You whack it what happens? Steps for solution:
 - o Define principal axes with a sensible origin.
 - \circ Calculate an expression for J in terms of the impulse:

$$\boldsymbol{J} = \int \boldsymbol{\tau} \, \mathrm{d}t = \int \boldsymbol{r}_{\scriptscriptstyle B} \times \boldsymbol{F} \, \mathrm{d}t = \boldsymbol{r}_{\scriptscriptstyle B} \times \int \boldsymbol{F} \, \mathrm{d}t = \boldsymbol{r}_{\scriptscriptstyle B} \times \boldsymbol{P}$$

Where B is the point at which the whack occurred, and r_B can be taken out of the integral because the whack is assumed to be instantaneous.

- $\circ~$ Work out an expression for \boldsymbol{J} in terms of $\boldsymbol{\omega},$ using the moments of inertia.
- \circ Equate the two expressions for J.
- o Work out the motion of the CM using standard linear mechanics.
- Note: The obvious origin to use is the CM, but other origins can be used subject to the provisos above for using $\tau = \dot{J}$. So a pivot, for example, is fine to use.
- You apply a torque what's the frequency of rotation?
 - Define principal axes with a sensible origin (eg: the CM see above).
 - \circ Find an expression for ω in these axes (with unknown magnitude), and find a corresponding expression for J, using the principal moments of inertia.
 - Find $d\boldsymbol{L} / dt = \boldsymbol{\omega} \times \boldsymbol{L}$.
 - $\circ \quad \text{Calculate the torque} \ (= \ \boldsymbol{r} \times \boldsymbol{F} \) \ \text{and equate it with} \ \mathrm{d} \boldsymbol{L} \, / \, \mathrm{d} t \, .$

Free Motion – Euler's Equation

- Free precession is a situation in which F = 0 and G = 0. In such a case, J is constant. ω is constant if J is along one of the principal axes, but otherwise, it will change direction, and perhaps even magnitude.
- We use the **Euler Equations** to analyse this problem.

$$\left[\frac{\mathrm{d}\boldsymbol{J}}{\mathrm{d}t}\right]_{\mathrm{lab}} = \left[\frac{\mathrm{d}\boldsymbol{J}}{\mathrm{d}t}\right]_{\mathrm{PA}} + \boldsymbol{\omega} \times \boldsymbol{J}$$

• Now, let's assume that a couple G is being applied in the **lab** frame. We know that

$$\boldsymbol{G} = \left[\frac{\mathrm{d} \boldsymbol{J}}{\mathrm{d} t} \right]_{\mathrm{lab}}$$

Therefore, using the equations above:

$$\boldsymbol{G} = \left[\frac{\mathrm{d} \boldsymbol{J}}{\mathrm{d} t} \right]_{\mathrm{PA}} + \boldsymbol{\omega} imes \boldsymbol{J}$$

• Finally, we note that in the principal axes frame, $\mathbf{J} = (I_1\omega_1, I_2\omega_2, I_3\omega_3)$. Therefore, casting **both sides** of this equation into the principal axes frame only

$$\tau_1 = I_1 \dot{\omega}_1 + (I_3 - I_2) \omega_3 \omega_2$$

And similarly with any ${\bf cyclic}\ {\bf permutation}$ of indices.

- A few notes:
 - All the quantities in this equation are measured with respect to the body frame (which is moving). This is the advantage of these equations – all we have to consider is the forces that the body "feels".
 - The two terms of the RHS refer to two types of ways J can change because it can change in the body frame and also because the body frame is itself rotating.

Free Motion – **Examples**

- FREE SYMMETRIC TOP
 - For a symmetrical top $(I_1 = I_2 = I)$ which is free in space (ie: no torque) the Euler Equations become

$$\begin{split} I\dot{\omega}_1 + (I_3 - I)\omega_3\omega_2 &= 0\\ I\dot{\omega}_2 + (I - I_3)\omega_1\omega_3 &= 0\\ I_3\dot{\omega}_3 &= 0 \end{split}$$

o The last equation implies ω_3 is **constant**. Let's define

$$\Omega = \frac{I_3 - I}{I} \omega_3$$

Then the general solution of the first two equations becomes:

$$\begin{pmatrix} \omega_1 \\ \omega_2 \end{pmatrix} = A \begin{pmatrix} \cos[\Omega t + \phi] \\ \cos[\Omega t + \phi] \end{pmatrix}$$

• Interpretation from the body frame

- In the body frame, ω₁ and ω₂ seem to form a circle in the x-y plane, with frequency Ω. How high that circle is depends on ω₃.
- L could be above ω (if I₃ > I an oblate top) or below
 ω (if I₃ < I a prolate top).

• Interpretation from the fixed lab frame

• In that case, the Euler Equations are useless, because they deal with the body frame, so we express things from scratch, but **in terms of the body frames**:

With Ω defined as above.

- This linear relationship between ω , L and \hat{x}_3 implies that they are in the same plane.
- Furthermore, the rate of change of x̂₃ is ω×x̂₃, because it only changes as a result of the rotation. So

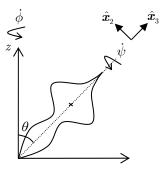
$$\frac{\mathrm{d}\hat{\boldsymbol{x}}_{3}}{\mathrm{d}t} = \left(\frac{\boldsymbol{L}}{I} - \Omega\hat{\boldsymbol{x}}_{3}\right) \times \hat{\boldsymbol{x}}_{3} = \left(\frac{\boldsymbol{L}}{I}\right) \times \hat{\boldsymbol{x}}_{3}$$

This is equivalent to \hat{x}_3 rotating at a frequency L/I.

• It turns out that we can interpret ω as follows

$$\boldsymbol{\omega} = \overbrace{\frac{I}{I}}^{\text{Motion of body around } L} - \overbrace{\Omega \hat{\boldsymbol{x}}_{3}}^{\text{Motion of body about}}$$

- HEAVY SYMMETRIC TOP
 - $\circ~$ Here, we must define the ${\bf Euler \ angles}$ as follows



• The total angular velocity is then given by

$$oldsymbol{\omega} = \widecheck{\dot{\psi} \hat{x}_3}^{ ext{Rotation of top}} + \widecheck{\dot{ heta} \hat{x}_1 + \dot{\phi} z}^{ ext{Motion of top itself}}$$

Which can be expressed in terms of the body-frames only:

$$oldsymbol{\omega} = \dot{\psi} \hat{oldsymbol{x}}_3 + \dot{ heta} \hat{oldsymbol{x}}_1 + \dot{\phi} \left(\hat{oldsymbol{x}}_3 \cos heta + \hat{oldsymbol{x}}_2 \sin heta
ight)$$

 $oldsymbol{\omega} = \left(\dot{\psi} + \dot{\phi} \cos heta
ight) \hat{oldsymbol{x}}_3 + \left(\dot{\phi} \sin heta
ight) \hat{oldsymbol{x}}_2 + \dot{ heta} \hat{oldsymbol{x}}_1$

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Dynamics – Normal Modes

Introduction

- A normal mode of a system is an oscillation that has a single frequency.
- All the more general oscillations of the system can be expressed as superpositions of these normal modes.

General approach

- Consider a system defined by generalised coordinates q_i and acted on by forces F_i , moving in a potential well $U(\mathbf{x})$, and moving elastically.
- The **kinetic energy**, T, is then given by

$$T = \frac{1}{2} \sum \sum m_i \left| \dot{\boldsymbol{\xi}}_j(q_i) \right|^2$$

Where $\sum_{i} \boldsymbol{\xi}_{j}(q_{i})$ is the **Cartesian coordinate** of the j^{th} part of the system, taken about an **equilibrium**, where all the $\boldsymbol{\xi}_{j}$ are 0. Expanding about that equilibrium:

$$\sum_{i} \boldsymbol{\xi}_{j}(q_{i}) = \sum_{i} \boldsymbol{\xi}_{j}(q_{i,\text{eq}}) + \frac{\partial \boldsymbol{\xi}_{j}}{\partial q_{i}} \bigg|_{\text{eq}} q_{i} + \cdots$$
$$\sum_{i} \dot{\boldsymbol{\xi}}_{j}(q_{i}) \approx \sum_{i} \frac{\partial \boldsymbol{\xi}_{j}}{\partial q_{i}} \bigg|_{\text{eq}} \dot{q}_{i}$$

And so:

$$T = \frac{1}{2} \sum \sum M_{ij} \dot{q}_i \dot{q}_j = \frac{1}{2} \dot{\boldsymbol{q}}^T \boldsymbol{M} \dot{\boldsymbol{q}}$$

Where

$$M_{ij} = \sum \sum m \frac{\partial \boldsymbol{r}}{\partial q_i} \bigg|_{\rm eq} \frac{\partial \boldsymbol{r}}{\partial q_j} \bigg|_{\rm eq}$$

Consider the potential energy, about a point of equilibrium (ie: a minimum in U) at which all the q_i are chosen to be 0.

$$U(x) = U_0 + \sum_{\substack{0 \text{ since at a minimum}}} \frac{\partial U}{\partial q_i} \Big|_{Eq} q_i + \sum_{\substack{0 \text{ since at a minimum}}} \frac{1}{2} \frac{\mathrm{d}^2 U}{\mathrm{d} x_j \mathrm{d} x_i} \Big|_{x_0} q_i q_j + \cdots$$
$$U(x) = U_0 + \frac{1}{2} \sum_{\substack{0 \text{ since at a minimum}}} K_{ij} q_i q_j + \cdots$$
$$U(x) = U_0 + \frac{1}{2} q^T K q$$

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• The **total energy** is then

$$E = U_0 + \frac{1}{2} \sum M_{ij} \dot{q}_i \dot{q}_j + \frac{1}{2} \sum K_{ij} q_i q_j$$
$$\frac{\mathrm{d}E}{\mathrm{d}t} = \frac{1}{2} \sum 2\dot{q}_i \left(M_{ij} \ddot{q}_j + K_{ij} q_j \right) = 0$$
$$\frac{\mathrm{d}E}{\mathrm{d}t} = \sum \dot{q}_i \left(M_{ij} \ddot{q}_j + K_{ij} q_j \right) = 0$$

• [Non rigorous argument] – the equations of motion are then:

$$\sum \sum M_{ij} \ddot{q}_j + \sum \sum K_{ij} q_j = 0$$
$$\boxed{\boldsymbol{M} \ddot{\boldsymbol{q}} + \boldsymbol{K} \boldsymbol{q} = 0}$$

• If we seek normal modes of the form $q(t) = Qe^{i\omega t}$, we get:

$$\left(\boldsymbol{K} - \omega^2 \boldsymbol{M}\right) \boldsymbol{Q} = 0$$

Non-trivial solutions only exist if

$$\det(\boldsymbol{K}-\omega^2\boldsymbol{M})=0$$

This defines the ω^2 normal mode frequencies.

- In **practice**, the steps are:
 - Find the K and M matrices by writing them out in terms of the variables of the system, and comparing with

$$T = \frac{1}{2} \dot{\boldsymbol{q}}^T \boldsymbol{M} \dot{\boldsymbol{q}}$$
 $U = U_0 + \frac{1}{2} \boldsymbol{q}^T \boldsymbol{K} \boldsymbol{q}$

Both matrices **must** be **symmetric**.

o Use the determinant method above.

Dynamics – **Elasticity**

Introduction

• Hooke's Law states that

$$\underbrace{\overbrace{F}}^{\text{Stress}} = E \, \underbrace{\frac{\Delta l}{l}}_{l}$$

Where

- \circ **F** is the force applied to a block of material over an area **A**.
- $\circ \Delta l$ is the extension of the block in the direction of F.
- \circ *l* is the **original**, **relaxed length** of the block *in that direction*.
- \circ *E* is the **Young's Modulus** of the material.
- Furthermore, it states that

$$\frac{\Delta w}{w} = -\sigma \frac{\Delta l}{l}$$

Where Δw is the length of the block in any direction perpendicular to that of l.

- For an isotropic material, E and σ are all we need to define the elastic properties of the material.
- Since these equations are all **linear**, the **principle of superposition** applies. If we have **several forces**, the **displacements** will be the **sum of** the displacements with the forces acting **individually**.

Uniform Strain – the Bulk Modulus

- Consider a **rectangular** block in a **pressure tank**, say, with **identical stress** *p* on every face.
- Consider one direction the change in length Δl in that direction is given by

$$\frac{\Delta l}{l} = \underbrace{\frac{-p}{E}}_{l} + \underbrace{\frac{\partial l}{\partial p}}_{E} + \frac{\partial l}{\partial p} + \frac{\partial l}{E} + \frac{\partial l}{\partial p} + \frac{\partial l}{$$

The problem is **symmetrical**, so the value will be the same for **all directions**.

$$\frac{\Delta V}{V} = \frac{\Delta x}{x} + \frac{\Delta y}{y} + \frac{\Delta z}{z}$$

We therefore have

$$\frac{\Delta V}{V} = -3\frac{1-2\sigma}{E}p$$

• We can then define the **bulk modulus**

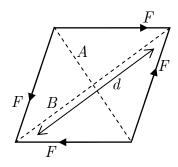
$$\boxed{K = \frac{E}{3(1 - 2\sigma)}}$$

Such that the **change of volume** as a result of the **stress** p is

$$p=-K\frac{\Delta V}{V}$$

Shear Strain – the Shear Modulus

• Consider a cube with face area A and with shear forces acting on it



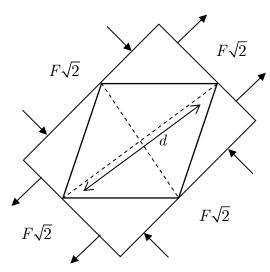
If cut the cube along the diagonals A and B, we find that

• There is a stretch normal to A, of magnitude $F\sqrt{2}$.

• There is a compression normal to B, of magnitude $F\sqrt{2}$.

And each of these diagonal faces has **area** $A\sqrt{2}$.

• The lengthening of the diagonal *d* will therefore be equal to the lengthening of *d* in the following case:

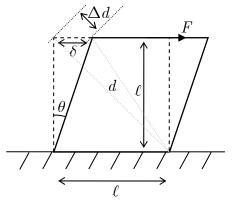


From above, this is given by:

$$\frac{\Delta d}{d} = \frac{1}{E} \frac{F\sqrt{2}}{A\sqrt{2}} + \sigma \frac{1}{E} \frac{F\sqrt{2}}{A\sqrt{2}}$$
$$\frac{\Delta d}{d} = \frac{1+\sigma}{E} \frac{F}{A}$$

By symmetry, the other diagonal is shortened by the same amount.

• It is often useful to have this as a function of the **twist angle**:



From this diagram, it is (reasonably) clear that

$$\delta = \Delta d\sqrt{2} \qquad \qquad d = \ell\sqrt{2}$$

Therefore

$$\theta \approx \frac{\delta}{l} = \frac{\Delta d\sqrt{2}}{l} = 2\frac{\Delta d}{d} = \frac{2(1+\sigma)}{E}\frac{F}{A}$$

• We therefore define the **shear modulus** as

$$\mu = \frac{E}{2(1+\sigma)}$$

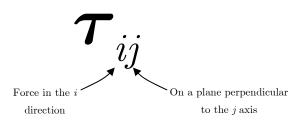
Such that

$$g = \mu \theta$$

Where g is the shear stress = F/A.

Formal Definitions

- Stress
 - Defined in terms of *force/unit area* transmitted across **planes** in the medium.
 - o Requires a **tensor**. We define



• We can then show that the force on **any arbitrary area element** is

$$F = \tau \,\mathrm{d}S$$

The tensor must be symmetric – consider a small cube side dx.
 Because the cube must be in equilibrium, the forces on it are as follows:

$$S_{yx}$$

The net **couple** on the cube is

$$\left(S_{xy}-S_{yx}
ight)\mathrm{d}x$$

But there must be **no torque** on the cube, or it'd spin! So

$$S_{\scriptscriptstyle xy}=S_{\scriptscriptstyle yx}$$

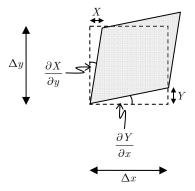
- o The stress tensor is diagonal for suitable choices of axes.
- The stress in a solid material is therefore described by a **tensor** field.
- Strain
 - When a material is put under strain, a point (x, y, z) in it is moved to a point (x + X, y + Y, z + Z).
 - The derivatives of these X, Y and Z contain information about the strain.
 - $\circ~$ As we saw before, it's worth considering two kinds of strain
 - For the **normal strains**, we define:

$$e_{xx} = \frac{\partial X}{\partial x}$$
 $e_{yy} = \frac{\partial Y}{\partial y}$ $e_{zz} = \frac{\partial Z}{\partial z}$

For example, if we consider stress **perpendicular** to the \boldsymbol{x} direction in a cube initially of side Δx , it'll increase by $e_{xx}\Delta x$:



• Now, for the **shear stresses**, consider

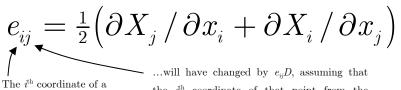


[The expression for the angles are tricky to see – but consider that X is the **change** in x...] We then simply define

$$e_{xy} = e_{yx} = \frac{1}{2} \left(\frac{\partial Y}{\partial x} + \frac{\partial X}{\partial y} \right)$$

This ensures that if the block simply *rotates* (ie: $\partial Y / \partial y = \partial X / \partial x$), these strains are 0.

o So in general, we define



point in the material...

...will have changed by $e_{ij}D$, assuming that the j^{th} coordinate of that point from the origin is D.

So, for example

$$X = e_{xx}x + e_{xy}y + e_{xz}z$$

- The tensor is also symmetric, due to the $e_{xy} = e_{yx}$ condition.
- o If the strains are **non-homogenous**, we sit down and cry.
- The relation between them

• Each component of e is related to *each* component of τ – this gives, overall, a fourth-rank tensor of elasticity relating the two:

$$\tau_{ij} = C_{ijkl} e_{kl}$$

(Using the summation convention).

- It looks like there are 9² = 81 coefficients in C, and that 81 numbers are therefore required to define the elastic properties of a material! However, we note that since S and e are symmetry, we must be able to swap ij and kl in C without changing a thing, so there can be at most 36 different coefficients.
- o If the material is isotropic, though, C must be completely frameindependent. As such, we must be able to express it in terms of the tensor δ_{ij}. There are only two ways of doing this that are also invariant under i ↔ j and l ↔ k, and so

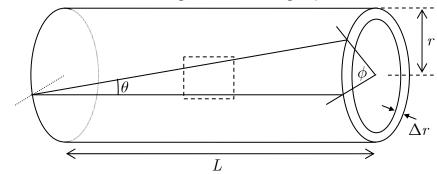
$$C_{ijkl} = \lambda \left(\delta_{ij} \delta_{kl} \right) + \mu \left(\delta_{ik} \delta_{jl} + \delta_{il} \delta_{jk} \right)$$

So an isotropic material only requires two constants (E and σ , for example). And we have

$$S_{ij} = \lambda e_{kk} \delta_{ij} + 2\mu e_{ij}$$

Examples – **Statics**

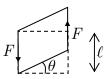
- Thin tube in torsion
 - o Consider a **thin tube** being **twisted** an angle ϕ



• We first note that

$$\theta = \frac{r\phi}{l}$$

• Next, consider a small square (dotted above) and its deformation as a result of the twist:



From the previous result:

$$\frac{F}{\ell\Delta r} = \mu\theta$$
$$F = \mu \frac{r\phi}{L} \ell\Delta r$$

• This force contributes a torque Δau to the rod

$$\Delta \tau = rF = \mu \frac{r^2 \phi}{L} \ell \Delta r$$

• Considering these bits around the whole rod, so that $\ell \to 2\pi r$, we get

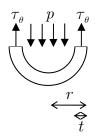
$$\tau = 2\pi\mu \frac{r^3 \Delta r}{L}\phi$$

• Wire in torsion

For a wire, we simply integrate the above from r = 0 to the total radius, giving

$$\tau = \mu \frac{\pi r^4}{2L} \phi$$

- Can under pressure
 - Consider a can of thickness t with closed ends with an internal pressure p.
 - Let the **tangential stress** in the walls be τ_{θ} , and consider *half* the can



The forces (= $stress \times area$) must balance, so

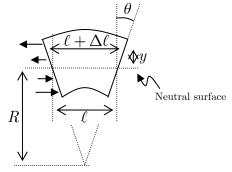
$$\begin{aligned} \tau_{\scriptscriptstyle\theta} \times 2t &= p \times 2r \\ \tau_{\scriptscriptstyle\theta} &= \frac{pr}{t} \end{aligned}$$

• Let the axial stress in the walls be σ_z , and consider one of the ends. By the same logic as above

$$\begin{split} \tau_z \times 2\pi rt &= p \times \pi r^2 \\ \tau_z &= \frac{pr}{2t} \end{split}$$

• Bent beam

- \circ Consider a **beam** of length L, held in a **bent** position.
- We only consider **longitudinal strains** (valid for **small deflections** and **thin beams**).
- Clearly, the bits at the top of the beam will be stretched, while those at the bottom will be compressed. Somewhere in between, there'll be a neutral surface – neither stretched nor compressed.
- o Consider a small segment length ℓ of the bent beam:



• The amount of stretching and compression at any point is proportional to the distance from the neutral surface, y. The constant of proportionality is ℓ/R . As such

$$\frac{\Delta \ell}{\ell} = \text{Strain} = \frac{y}{R}$$

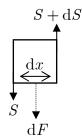
• Clearly, there'll be forces to the **left above** the neutral surface, and **vice versa**. We therefore have

$$\frac{\Delta F}{\Delta A} = E \frac{\Delta \ell}{\ell}$$
$$\Delta F = \frac{E}{R} y \Delta A$$

o The total torque produced about the neutral line is given by

$$\tau = \int_{\text{Cross}} y \, \mathrm{d}F$$
$$= \frac{E}{R} \int_{\text{cross}} y^2 \, \mathrm{d}A$$
$$\boxed{B = \frac{EI}{R}}$$

• Now, consider a **beam loaded** with **weights** given by W(x), where W is the **force per unit length**. Consider the statics of a small **segment** of the beam:



Notes:

• Due to the **bending moment**, some **vertical forces** are produced. Ignoring products of infinitesimal quantities, we can write, **at that point**

$$S \, \mathrm{d}x = \mathrm{d}B$$
$$B = \frac{EI}{B} = \int S \, \mathrm{d}x$$

[Effectively, we're saying that due to the dS needed to balance dF, the bending moment must change]

The downwards loading force needs to be balanced by a difference in the upwards stress

$$dS = dF = W \, dx$$
$$S' = W$$

 $\circ \quad {\rm Now, \ for \ small \ deflections}$

$$y'' = 1/R$$

o As such, we can conclude

$$EIy'''' = W(x)$$

- o **Boundary conditions** for various cases are as follows
 - At a free end, S and B are clearly 0, and so y'' = y''' = 0.
 - At a **cantilevered end**, y and y' are given (usually 0).
- Finding y is then simply a question of solving that differential equation. However, there are a few tricky points
 - All forces must be considered when writing down W(x), including reactions at contacts. Most often, W will be a series of δ -functions.
 - Sign conventions:

- Downwards $W \rightarrow$ positive.
- The resulting y obtained is **downwards** \rightarrow **positive**, because the way the radius of curvature is specified.
- However, be *very* careful sometimes, the convention appears to be reversed because the bar curves downwards, and so -1/R = y''.
- Don't worry too much about boundary conditions for y^{'''}

 just integrate δ-functions from 0 to L (for a free end, this is fully justified). Remember that there'll often be a δ-functions at the very end of the range, which might help satisfy the boundary conditions.
- From then on, **boundaries** are just provided. Just also remember to make the y'', y' and y continuous.
- The **couple** provided by a **cantilever** can simply be worked out by evaluating B = EIy'' at that point.
- It is sometimes easier to simply write down y'', the bending moment from physical considerations.
- $\circ~$ The Euler Strut is a beam buckled between two walls:

$$F$$
 F

If we take y upwards, then the bending moment on any point is B = -Fy

$$y'' = -\frac{F}{EI}y$$
$$y = A\sin\left[x\sqrt{\frac{F}{EI}}\right]$$

Applying the boundary condition that y = 0 at x = L:

$$F = \frac{\pi^2 EI}{L^2}$$

This is **independent of displacement** (but only while y'' = 1/R holds).

o The Reciprocity Theorem states that

"The deflection at Q due to a load at P is the same as the deflection at P due to the same load at Q" To prove, say P_{PQ} means "the deflection at P due to the load at Q". Consider loading first P and then Q. The energy stored is

$$E = F \left[\frac{P_{PP}}{2} + \frac{P_{QQ}}{2} + P_{PQ} \right]$$

The same result must be applied the other way round, so

$$P_{PQ} = P_{QP}$$

Dynamics of Rigid Bodies

• Consider a small volume V of the material. It will have both external forces acting on it (eg: gravity) and internal forces (eg: elastic stresses).

$$\boldsymbol{F}_{\mathrm{ext}} + \boldsymbol{F}_{\mathrm{int}} = \int \rho \ddot{\boldsymbol{r}} \,\mathrm{d}\,V$$

• Every small particle in the volume experiences the **external** force, though, so F_{ext} is given by a **volume integral**.

$$oldsymbol{F}_{ ext{int}} = \int \widecheck{(-oldsymbol{f}_{ ext{ext}} +
hooldsymbol{ec{r}})}_V \, \mathrm{d}\,V$$
 $oldsymbol{F}_{ ext{int}} = \int_V oldsymbol{f} \, \mathrm{d}\,V$

On the other hand, only the particles at the **edge** of the volume experience the **elastic** force from surrounding media, and so F_{int} is given by an **area integral**

$$\int_{\mathbf{A}} \boldsymbol{f}_{\text{int}} \, \mathrm{d}\boldsymbol{A} = \int_{V} \boldsymbol{f} \, \mathrm{d}\, V$$

• We have, however, defined that the force in the *x*-direction, say, is

$$\mathrm{d}F_x = \left(S_{xx}\boldsymbol{i} + S_{xy}\boldsymbol{j} + S_{xz}\boldsymbol{k}\right)\cdot\mathrm{d}\boldsymbol{A}$$

And so, taking only the x component of the integral above

$$\int_{\mathbf{A}} \left(S_{xx} \boldsymbol{i} + S_{xy} \boldsymbol{j} + S_{xz} \boldsymbol{k} \right) \cdot \mathrm{d} \boldsymbol{A} = \int_{V} f_{x} \, \mathrm{d} V$$

• Using the Divergence Theorem on the LHS

$$\int_{V} \left(\frac{\partial S_{xx}}{\partial x} + \frac{\partial S_{xy}}{\partial y} + \frac{\partial S_{xz}}{\partial z} \right) \mathrm{d} V = \int_{V} f_{x} \, \mathrm{d} V$$

Removing the volume integrals (because this is true for any volume):

$$f_i = \partial S_{ij} / \partial x_j$$

(Using the summation convention).

• Now, using $\tau_{ij} = \lambda e_{kk} \delta_{ij} + 2\mu e_{ij}$ (isotropic material), we obtain

$$\boldsymbol{f} = (\lambda + \mu) \nabla (\nabla \cdot \boldsymbol{u}) + \mu \nabla^2 \boldsymbol{u}$$

Where u is the internal displacement in the solid.